

BOOK41R

Notes:

"U-Teflon 1966" on spine

Blank pages: inside front cover sheets, 1-4, 6, 8-10, 12, 14, 16, 18, 20, 24, 26, 28, 30, 38-40, 42-50, 52, 58-60, 62, 68, 70, 74, 78, 79, 112, 114, 116-308, inside back cover sheets

- page 9 has 1 (8.5x11) sheet taped
- page 18 has 1 (8.5x11) sheet taped
- page 31 has 1 graph glued
- page 36 has 2 graphs glued
- page 37 has 1 sheet of paper glued
- page 72 has 1 sheet of paper glued
- page 96 has 1 graph sheet glued/taped
- page 97 has 1 graph sheet glued
- page 101 has 1 graph sheet glued
- page 110 has 1 sheet taped

Scanned by:

Sheila Finch

RSICC /Oak Ridge National Lab.

August 10, 1999



PIONEERS SINCE 1831

Account Book

No. S 149

NO UNITS

Journal

Ledger, Single Entry . .

Ledger, Double Entry .

Record Ruled (27 Lines)

Made in 150, and 300 Pages

MADE IN U. S. A.

TO REORDER, SPECIFY NUMBER,
RULING AND THICKNESS INDICATED
ON BACKBONE OF THIS BOOK.

- September 23 1966 Completed checking experiments at 37.5% and 4.89%.
- Oct 7, Calculations (S_2) completed.
- Oct 10-13 Int. Conf on Fast Critical Experiments, ANL
 Talked with C.E. COHN re Computers in Crit Exp.
 Reactivity meas and Rod Calibration using digital code similar to RTF code (Lozlanos)
 Noise analysis, various to mean etc.
 Ion chamber current converted to Fm signal to improve statistics of counting.
 Talked with E Bennett re: Neutron energy spectra with H recoil prop. counters.
- Oct 14 Zedler - discussed H recoil prop counter, referred me to Verbinski who will loan us one which he had made for testing so called Mark O
 0.002" inside SS, 17/16" cathode brass
 150 cc Hg, CH_4
- | Volts | A |
|-------|------|
| 2000 | 1.6 |
| 2500 | 3.6 |
| 3000 | 16. |
| 3500 | 86. |
| 4000 | 530. |
- counter not designed but built for testing.
- 17 Verbinski - counter + copies of response & mentions on polaroid camera photo of PHS data. & Zedler's amp. curve.

Nov 31, 1966

After discussions with ADC, looked into low moderation experiments with layers of C, CF₂ and CH₂ in fully enriched metal systems. Ideal situation would be homogeneous CH₂ (paraffin) - UO₂ (or UO₃ or UF₄) blocks but cost may be prohibitive, est. to be 40-50 k\$. JTT has considerable supply of 7/8 X 5 X 10 metal pieces. Layers of 7/8 X 10 X 10 could be used.

(See 7/8 X 5 X 10 on split table apparatus)

A "Gedanken" experiment, calculated using DTF code to compare the homogeneity to the heterogeneity of 7/8 in. layers of C, CF₂ and CH₂ in 7/8 layers of U metal.

Metal 7/8 in = .875 = 2.22 cm

10 in x 10 in = 25.4 x 25.4 cm

NSE 8,575 h = 7.73 cm λ_s = 2.15
 d = 35.9 cm r = 17.95 cm λ_c = 2.10 cm

$$B^2 = \frac{\pi^2}{(11.93)^2} + \frac{2.405^2}{20.05^2} = .26334^2 + .11995^2$$

$$B^2 = .069348 + .014388 = .083736$$

For Sphere B = .2894 R̄ = 10.86 R = 8.71 cm

$$\frac{4}{3}\pi R^3 \rho = \frac{4}{3}\pi (8.71)^3 \rho = 51.87 \frac{kg}{cm^3} U @ 932 = 48.34 kg$$

$$B_x^2 = B_y^2 = \left(\frac{\pi}{29.6}\right)^2 = .010614^2 = 0.0112668$$

$$B_z^2 = .061204 B_2 = .2474 Z = 12.70$$

Vol = 5483.86 ← Z = 8.50 cm $\frac{Z}{2} = 4.25 cm$
 $^{235}U \times 17.466 = 95.78 kg$

$$\frac{8.5}{2.22} = 3.83 \text{ layers}$$

4.440 cm k = .9934
 4.218 k = .9663

Machined Slabs, 18.72 g/cc

93.470

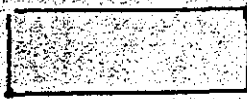
235 U

Length
Width

5" x 10"

Size
(Thickness)

$\frac{7}{8}$ and $\frac{1}{8}$

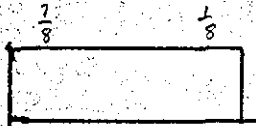


Amt. 10 ea.

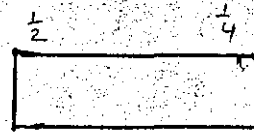
$\frac{1}{2}$ and $\frac{1}{4}$



2 ea.

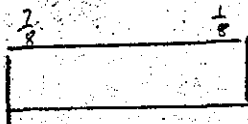


4 ea.

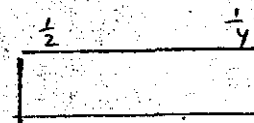


2 ea.

3" x 10"

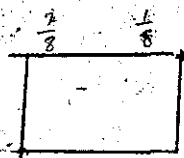


4 ea.

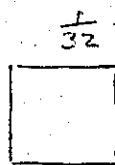


2 ea.

2" x 10"



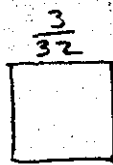
4 ea.



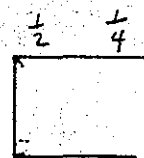
2 ea



25 ea

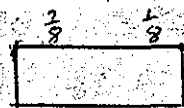


25 ea

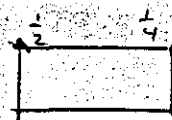


4 ea.

5" x 5"

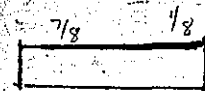


4 ea.

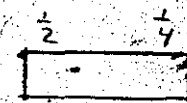


4 ea.

2" x 5"

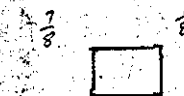


2 ea



2 ea.

1" x 5"



4 ea.

1" x 1"

Dr

Metal

$$93.4 \text{ wt\% } ^{235}\text{U} \times 18.74 = 17.50 \text{ g } ^{235}\text{U/cm}^3$$

$$6.6 \text{ wt\% } ^{238}\text{U} = 1.24 \text{ g } ^{238}\text{U/cm}^3$$

$$\frac{17.50 \times 0.60228 \times 10^{24}}{235.05} = 0.04484 \times 10^{24}$$

$$\frac{1.24 \times 0.60228}{238.06} = 0.00314 \times 10^{24}$$

Graphite

$$1.7 \text{ g/cm}^3$$

$$\frac{1.7 \times 0.60228}{12} = 0.08532 \times 10^{24}$$

CF_2

$$2.17 \text{ g/cm}^3$$

$$\text{C} = 0.24 \times 2.17 = 0.5208 \quad \text{F} = 0.76 \times 2.17 = 1.6492$$

$$\text{C} = \frac{0.5208 \times 0.60225}{12} = 0.02614 \times 10^{24}$$

$$\text{F} = \frac{1.6492 \times 0.60225}{19} = 0.05228 \times 10^{24}$$

CH_2

$$0.92 \text{ g/cm} \quad \text{H} = \frac{1}{2} \times 0.92 = 0.46$$

$$\text{C} = 0.46$$

$$\text{C} = \frac{0.46 \times 0.60228}{12} = 0.02311 \times 10^{24}$$

$$\text{H} = \frac{0.46 \times 0.60228}{1} = 0.27704 \times 10^{24}$$

Answer

OTF Calculations - Sy Approx.

$$B_x^2 = B_y^2 = 0.011266$$

Half Slab Calculation in Z direction

Mesh Spacing 0.222 cm

Case A) 20 Mesh Spaces 4.44 cm half thickness
HOMOGENEOUS $^{235}U = 4.484 \text{ E-2}$

~~Case B~~

$$^{238}U = 3.140 \text{ E-3}$$

Case B) 19 Mesh Spaces 4.218 cm half thickness

CASE B REGION 1, 5 Mesh Spaces (1.11 cm)

$$\text{Material 1 } ^{235}U = 4.484 \text{ E-2}$$

$$^{238}U = 3.140 \text{ E-3}$$

Region 2, 10 Mesh Spaces (2.22 cm)

$$\text{Material 2 } C = 8.532 \text{ E-2}$$

Region 3, 10 Mesh Spaces (2.22 cm)

Mat'l 1

Region 4, 10 Mesh Spaces (2.22 cm)

Mat'l 2

Region 5, 5 Mesh Spaces (1.11 cm)

Mat'l 1

CASE C HOMOGENEOUS 40 Mesh Spaces

(8.88 cm)

$$^{235}U = 2.242 \text{ E-2}$$

$$^{238}U = 1.570 \text{ E-3}$$

$$C = 4.266 \text{ E-2}$$

CASE D (Same as B except for Mat'l 2) CF_2
 Material 2D = C = 2.614 E-02
 F = 5.228 E-02

CASE E HOMOGENEOUS
 ^{235}U 2.242 E-02
 ^{238}U 1.570 E-03
 C 1.307 E-02
 F 2.614 E-02

CASE F (Same as B except for Mat'l 2) CH_2
 Mat'l 2 F C = 3.957 E-02
 H = 7.914 E-02

CASE G HOMOGENEOUS
 ^{235}U 2.2420 E-02
 ^{238}U 1.5700 E-03
 C 1.9785 E-02
 H 3.9570 E-02

1:1 UCF6-CH2 Blocks

12 x 12 x 12 (-)

1688 blocks (vs 1728)

(actual vol. $\sim 1.5\%$ greater)
 $\rightarrow 27661 \text{ cm}^3$

$$B^2 = 3 \times \frac{\pi^2}{(30.5+5.0)^2} = 3 \times (0.0885)^2 \quad S = 2.5$$

$$= 2.3496 \times 10^{-2}$$

$$\tilde{R} = \frac{\pi}{S} = \frac{\pi}{2.5} = 20.49 = 20.5$$

$$B = 1.533 \times 10^{-1}$$

$$R = 18.0 \text{ cm}$$

Atom Density $\times 10^{-24}$

$$M_c = 41.2 \text{ kg } ^{235}\text{U} \rightarrow 105.57 \times 10^{24} / 27661 = 3.8166 \text{ E-03}$$

$$2.0 \text{ kg } ^{238}\text{U} \quad 5.06 \times 10^{24} / 27661 = 1.829 \text{ E-04}$$

$$\text{Total U} = 3.9995 \text{ E-03}$$

C

$$3.9995 \text{ E-03}$$

F (x6)

$$23.997 \text{ E-03}$$

C H_{1.92} blocks density of 0.91

$$2.09 \text{ gm H} / 16.387 = .12754$$

$$7.6815 \text{ E-02}$$

$$15.11 - 2.09 = 13.02 \text{ gm C} = .79453$$

$$3.9844 \text{ E-02}$$

12.01

16.387

Total C

$$4.3844 \text{ E-02}$$

Block volumes are $\sim 1\frac{1}{2}\%$ greater than nom -

Assume actual stacking densities are less by 0.5%

\therefore All densities above reduced 2%, use 98% of

theoretical

$$^{235}\text{U} \quad 3.7403 \text{ E-03}$$

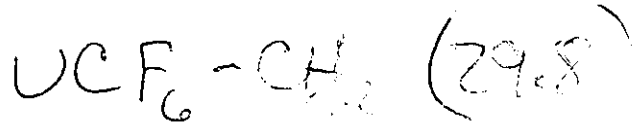
$$^{238}\text{U} \quad 1.7920 \text{ E-04}$$

$$\text{C} \quad \frac{2.3444}{\cancel{2.3444}} \text{ E-02}$$

$$\text{F} \quad 2.3517 \text{ E-02}$$

$$\text{H} \quad \frac{3.7039}{\cancel{3.7039}} \text{ E-02}$$

$$\text{H}/\text{U}^{235} = \frac{10.063}{\cancel{10.063}}$$



16.584
16.632
07
16.612
16.630

1.015
1.005

$$\begin{array}{r}
 16.584 \\
 16.632 \\
 .07 \\
 \hline
 16.70 \\
 16.387 \\
 \hline
 1.0065^2 \\
 = 1.019 = \frac{16.7}{16.387}
 \end{array}$$

		$\omega +$ (g)	$\frac{1}{4.70} \cdot 0.05988$	$\frac{1.00278}{A^{1.019}}$							
235	U	15.38	.9210	2.3598	E-03		1.5732	E-03	1.1799	E-03	
238	U	36.18	2.1664	5.4809	E-03		3.6540	E-03	2.7404	E-03	
	U	51.56	3.0874	<							
	F	24.69	1.4784	4.6863	E-02		3.1242	E-02	2.3431	E-02	
	C	2.64	0.1581	0.7928	E-02		.5285				
	O	0.08	.0048								
	AL	0.56	.0335								
prob.	C	13.02	.7796	3.9096	E-02		1.3032				
	H	2.09	.1251	7.5345	E-02		2.5115	E-02	3.7672	E-02	
							C T T	1.8317	E-02	2.3512	E-02
							$\frac{H}{x} =$	15.96		31.93	

B^2 for $16 \times 16 \times 16 + 121$

for $14 \times 13 \times 13 + 98$

$16 \times 16 = 256$	$\times 2.50$	$+ 5.00$	$(14 \times 13) = 182$
Assume size = $1.0065 \times 16 = 16.104$	40.904	45.90	
$\times 16.5 = 16.607$	42.182	47.18	
$\times 14$	14.091	35.79	40.79
$\times 13$	13.085	33.24	38.24
$\times 13\frac{1}{2}$	13.588	34.51	39.51

$B^2 = \frac{\pi^2}{45.90} (6.844 \times 10^{-3})^2 = 4.684 \times 10^{-3}$

$B^2 = \frac{\pi^2}{40.79} (7.701)^2 = 5.931$

$\frac{\pi^2}{47.18} = 6.659$
 4.434

$\frac{\pi^2}{38.24} = 8.215$
 6.749
 $\frac{\pi^2}{39.51} = 7.951$
 6.322

$B^2 = 0.13802 \times 10^{-3}$

19.002×10^{-3}

$B = .1175$

$B = 1.379 \times 10^{-1}$

$\tilde{R} = 26.74$

$\tilde{R} = 22.78$

1:2 UCF₆-CH₂ BLOCKS (95%)

		66667	x	²³⁵ U	3.7403	E-03	2.4935	E-02	
				²³⁸ U	1.7929	E-04	1.1917	E-04	
			x 98	C	3.9495	E-03			
				F	2.8517	E-02	6.5678	E-02	
							2.6032		
	k	1.3333	x 98	C	1.9422	E-02			
1.015092	16.8 cm			H	3.7439	E-02	5.0185	E-02	
1.005183	16.6						Total C	2.3645	E-02

2:1 UCF₆-CH₂ (29.8%) 1:1 UCF₆-CH₂ (29.8%)

²³⁵ U	1.5732	E-03	1.1799	E-03
²³⁸ U	3.6540	E-03	2.7404	E-03
F	3.1242	E-02	2.3431	E-02
H	2.5115	E-02	3.7672	E-02
C	1.8317	E-02	2.3512	E-02

$\frac{H}{K} = 15.96$

Buckling set to 1.3802×10^{-2}

Sphere Radius = 24.24

R (cm) assume $\lambda_c = 2.5$ cm

768	24.24
138	24.00

Buckling 1.9002×10^{-2}

Sphere Rad = 20.28

k	R
433 481	20.28
922 495	20.28

Interpretation of NSE & 588 (Dec 1960)

~~Hansen Wood Bygones Osborne~~ Critical Masses

Hansen Wood & Beer, Crit Masses of Enriched Uranium Cylinders
with Multiple Reflectors.

H:U = 2.59 polyethylene 15 in o.d x 0.125 in, ~6.5 kg each
100% Ni end reflectors C.M. = 76.3 kg U
SS " " 79.6 "
Fe " " 82.0 "

H:U = 4.88 polyethylene 15 in o.d.
100% Ni end Refl. CM = 59.4 kg U
SS " " 62.0 "
Fe " " 62.5 "

H:U = 0 Hansen Paxton Wood 15.0 in o.d 3.25 in high
CM = 165.7 kg U

$$\frac{3.25}{.125} = 26 \text{ plates } \quad \frac{165.7}{26} = 6.373 \text{ kg/plate}$$

Stated plates were 0.120 weighing ~6.5 kg

$$\frac{3.25}{.12} = 27.08 \text{ plates} \times 6.5 = 176.0 \text{ kg}$$

$$\text{or } \frac{165.7}{27.08} = 6.123 \text{ kg/plate}$$

Something is not understood.

$$15 \text{ in o.d} \times 0.125 \text{ in} \approx \pi \frac{d^2}{4} L = 22,089 \text{ in}^3 \rightarrow 361.97 \text{ cm}^3$$

$$6.5 / 361.97 = 17.96 \text{ g/cm}^3$$

$$93.2 \text{ wt\% } \begin{array}{l} {}^{235}\text{U} = 16.74 \\ {}^{238}\text{U} = 1.22 \end{array} \quad \begin{array}{l} 4,289 - 02 \\ \underline{.309} \end{array} \quad 10^{-24}$$

$$4.598 - 02$$

$$\text{godiva mass corrected to } 17.96 \text{ density} = \left(\frac{18.8}{17.96}\right)^2 = 1.0958 \times 48i$$
$$= 52.6$$

H:U = 2.59 $\frac{76.3}{6.5} = 11.74$ $\frac{79.6}{6.5} = 12.25$ $\frac{820}{6.5} = 12.62$
~~H:U = 4.88~~ $\frac{59.4}{6.5} = 9.14$ $\frac{62.0}{6.5} = 9.54$ $\frac{62.5}{6.5} = 9.62$
~~H:U = 2.59~~

$2.59 \times 4.598 = 11.91 - 02$ Poly ethylene H density = $\frac{7.90 - 02}{1.51} = 5.23$
 $4.88 \times 4.598 = 22.44 - 02$ → $\frac{1.51}{3.075} \text{ cm CH}_2/\text{cm U}$
 → $2.84 \text{ cm CH}_2/\text{cm U}$

$.125 \text{ cm U} = .3175 \times 1.51 = .479 \text{ cm} \quad (.189 \text{ in})$
 $\times 2.84 = .902 \text{ cm} \quad (.355 \text{ in})$

Ni $11.74 \times (.3175 + .479) = 9.35 \text{ cm}$
 SS $12.25 \times .7965 = 9.76 \text{ ''}$
 Fe $12.62 \times 1.005 = 10.05 \text{ ''}$
 15.0 → 38.1m

2.1 one end only
 ESTIMATE REFLECTOR SAVINGS
~~19.50 - 7.45 = 12.05~~
 $11.50 + (\delta = 2.10)$
 $2\delta = 4.2$

Ni $9.14 \times (.3175 + .902) = 11.15 \text{ cm}$
 SS $9.54 \times 1.2195 = 11.63$
 Fe $9.62 \times 1.005 = 11.73$

$13.30 + (\delta = 2.10)$
 $2\delta = 4.2$

$B^2 = \frac{\pi \cdot 2.405^2}{(19.05 + 2.1)^2} + \frac{\pi^2}{15.7^2} = .1137^2 + .2001^2 = .05293$
 $B = .2301$

$B^2 = \frac{\pi^2}{17.5^2} = .01293 + .1795^2 = .04515$
 $B = .2125$

$\frac{4\pi R^3}{3} = 6.3706 \text{ L} \times 17.96 \times \frac{7.159}{7965} = 45.61 \text{ kg U}$
 vol fract .3986 @ 2.59
 $= 8.439 \text{ L} \times 17.96 \times \frac{4.682}{1.2195} = 39.5 \text{ kg U}$
 vol fract .261

$\tilde{R} = \frac{13.65}{2.15} = 6.35$
 $\tilde{R} = \frac{11.5}{2.15} = 5.35$
 $\tilde{R} = 14.78$
 12.63

~ Jan 20-24.

ANISN Calculations of UCF_6 (37) show flux in groups 10-7 approx. $\propto E$ and not $1/E$ as H.R. x sections are weighted. Calculations were made for estimating the errors involved in using $1/E$ weighting for $1/v$ x sections instead of constant flux or flux prop. to E .

$$\langle \sigma(1/v) \rangle_c / \sigma(E_1) = 2 \left[(E_2/E_1)^{1/2} - 1 \right] / \left[(E_2/E_1) - 1 \right] \quad \text{const } Q$$

$$= .714 \quad \text{for } E_2/E_1 = 3.24$$

$$\langle \sigma(1/v) \rangle_{1/E} / \sigma(E_1) = 2 \left[1 - (E_2/E_1)^{-1/2} \right] / \ln(E_2/E_1) \quad Q \propto \frac{1}{E}$$

$$= .756 \quad \text{for } E_2/E_1 = 3.24$$

$$\langle \sigma(1/v) \rangle_E / \sigma(E_1) = 4 \left[(E_2/E_1)^{3/2} - 1 \right] / \left[(E_2/E_1)^2 - 1 \right] \quad Q \propto E$$

$$= .678 \quad \text{for } E_2/E_1 = 3.24$$

Jan 31

ANISN calc. of UCF_6 with Rev F x section and $\sigma_p(238) \approx 89$ and $\sigma_p(235) = 54$ using mixture of 238-2R-3R and 235 3R-4R rather than original H.R. x sections. $k_{eff} = 1.037668$ S_4 43.5

$$- \frac{.001810}{1.035858} \quad S_8 \text{ corr}$$

$$1.035858$$

$$- \frac{.00861}{1.027248} \quad 43.0 \text{ rad}$$

$$1.027248$$

$$+ \frac{.00593}{1.033178} \quad \# \text{ correct}$$

$$\text{Best value is now } \underline{1.033178}$$

Errors due to incorrect flux weighting may be as large as 20% of the absorption in groups 9-10-11-12. and perhaps 10% in groups 5-6-7-8.

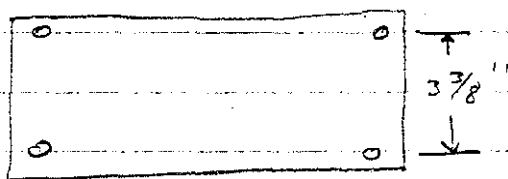
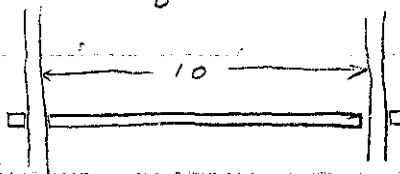
Febr 2

Design of supports for low mod exp.

10x10 slab \rightarrow 3.32 m hi \rightarrow 95.1 kg ^{235}U

5x10x3.32 \sim 50 kg

Use "tanktop" supports with $1/16 \times 5 \times 10$ base plate suggested by Tunnel.



Febr 7

Calculation concerned with safety of

stacking - 10x10x3.32 crit J T M

5x5x9.13 crit J T M

with 1" plexiglas 10x10x2.32 crit J T M

Assumed reflector savings = 0.5cm

Buckling for 2.1 cm exhaustion dist

Height	B^2	Configuration	B^2
5"	.03456	10x10x3.33	.08437
10"	.01126	5x5x9.13	.08228
3.32"	.06185	5x10x4.32	.08854
9.13"	.01316	= 5x10x3.32	
4.32"	.04272	plus reflector.	

Conclusion - no problem with hand stacking, 5x10x3.32

August 23, 1967

Idea! Suggestion!

Measurement of ^{238}U resonance integrals as a function of σ_s , scattering cross-section per ^{238}U atom. Use a poisoned flux trap for the γ E spectrum source with a central region which will contain solutions or mixtures of ^{238}U with scattering materials. Some boron or cadmium or suitable poison could be added to the uranium to ^{eliminate thermal capture in uranium} reduce thermal capture.

Range of interest for σ_s values would be 100 to 100 000

For $\text{UO}_2(\text{NO}_3)_2$ solutions 935 g/liter corresponds to H: ^{238}U ratio of 20 and $\sigma_s = 400$ -- impracticable

374 g/liter $\rightarrow 62 \rightarrow 1240$ b

3.74 g/liter $\rightarrow 7000 \rightarrow 139\ 000$ b

> primary
range of interest

A wide variety of enrichments and moderation values for many chemical compounds could be investigated.

Determine enrichment for which $k_{\infty} = 1$

1. for uranium compounds i.e. U metal, UF_6 , UO_2 , UC, UCF_6 , U_3O_8 and other compounds and alloys.
2. for uranium compounds with various amounts of hydrogen and other moderators
3. for uranium compound with various diluents offering high absorption cross section
4. for various combinations of 1, 2 & 3.
5. for various configurations for fast reactors.

A simplified description of the theory of the k_{∞} measurement
 For the reactor with a central void at delayed critical

$$k_1 = \frac{\text{production}}{\text{absorption} + \text{leakage}} = \frac{P_n}{A_n + L_n} \equiv 1$$

$$P_n = A_n + L_n$$

With sample having $k_{\infty} \neq 1$ in general in void

$$k_{\infty} = P_s / A_s$$

$$k_2 = \frac{\text{production (reactor + sample)}}{\text{absorption (reactor + sample) + leakage (reactor + sample)}} \\ = \frac{P_n + P_s}{A_n + L_n + A_s + L_s}$$

$$\Delta k = k_2 - k_1 = \frac{P_n + P_s - (A_n + L_n + A_s + L_s)}{A_n + L_n + A_s + L_s} \\ = \frac{P_s - A_s - L_s}{P_n + A_s + L_s} = \frac{P_s/A_s - 1 - L_s/A_s}{P_n/A_s + 1 + L_s/A_s} \\ = \frac{k_{\infty} - 1 - L_s/A_s}{\frac{P_n}{P_s} k_{\infty} + 1 + L_s/A_s}$$

$$L_s/A_s \ll 1 \ll P_n k_{\infty}/P_s$$

$$\Delta k \approx \frac{k_{\infty} - 1 - L_s/A_s}{\frac{P_n}{P_s} k_{\infty} + 1}$$

The leakage L_s can be made small by flux flattening techniques, or can be evaluated from flux traverse.

If typically, 1% of the fissions occur in the sample then the minimum detectable $(k_{\infty} - 1)$ would be (100 x) the minimum reactivity change detectable. The latter can be reduced to ≈ 0.2 cent. $(k_{\infty} - 1)_{\min} = 20 \text{ cent} = 0.0013$. $k_{\infty} = 1.0 \pm 0.0013$, for $k_{\infty} \neq 1$, the errors will be larger because of the difficulty in measuring reactivities with precision. (Also the ± 0.2 cent can be reduced near critical with sample oscillation techniques.)

Suggested calculations:

B. Spherical shells of

- $r_1 = 4 \text{ cm}$ 1. void and sample, k_1 and k_2 (10 mesh)
 $r_2 = 12 \text{ cm}$ 2. buffer region of sample (10 mesh)
 $r_3 = 14.5 \text{ cm}$ 4. enriched uranium metal shell (10 mesh)
 $r_4 = 19.5 \text{ cm}$ 5. steel reflector to reduce leakage (10 mesh)
 20.5 cm
 20.5 cm
 21.0 cm and for control region

Calculate k_1 and k_2 for best guess of enrichment of metal
 note flux in sample region, flatness of spectrum
 note flux in buffer region, " " "

A. k_{∞} calculations for ^{various} enrichments to
 find $k_{\infty}(\text{calc}) = 1$ for U-metal and
 other compounds, alloys, etc.

ESTIMATE OF CRITICAL SHELL THICKNESS.

Critical spherical shell thickness
 will be greater than $1/2$ the thickness of an ∞
 critical slab.

$$t(\infty \text{ slab}) = 6.66 \text{ cm}$$

$$\therefore t/2 = 3.33 \text{ cm}$$

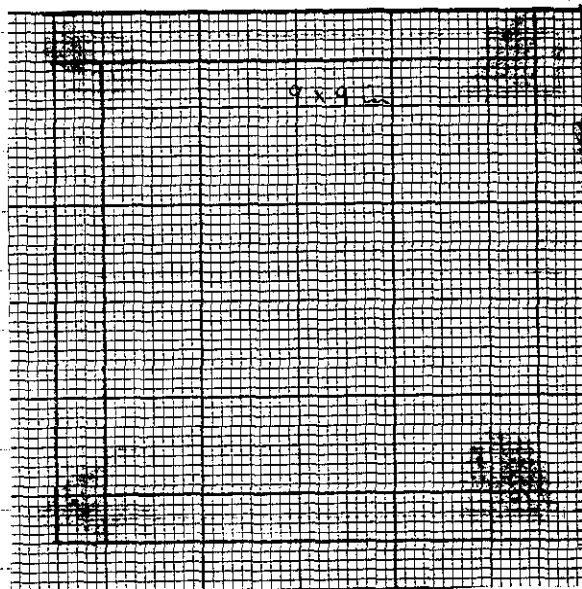
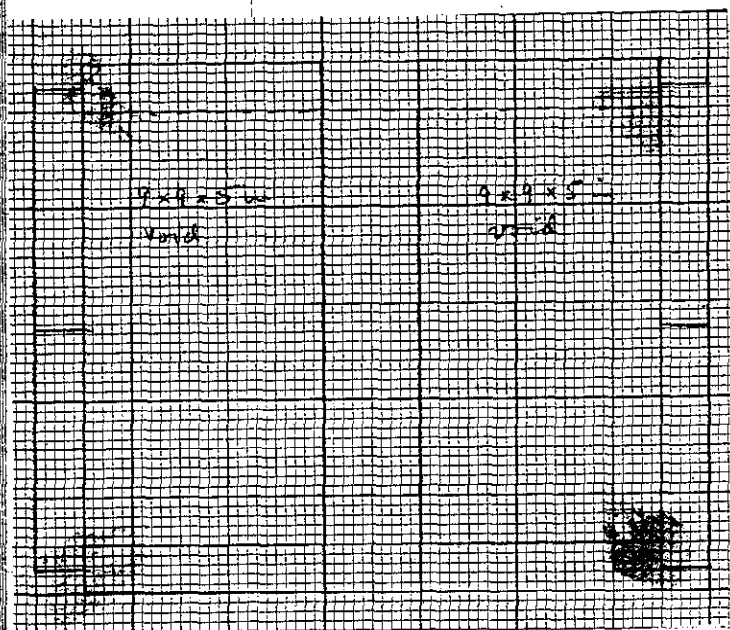
\therefore 1" thick spherical shell can be made critical
 with suitable reflector thickness. ∞ Steel reflector ~~thickness~~
 savings $\approx 1.5 - 1.8 \text{ cm}$ ($\infty \text{ H}_2\text{O} \sim 2.1 \text{ cm}$)

It should be possible to construct a rectangular, hollow paralleliped from the uranium slabs, 5x10 and 5x5 having various thicknesses. Inside dimensions 9x9x10 outside dimensions 11x11x12 except for some corners missing, using 9 ea 3x3 \square alt tubing 5 in long on each half of assembly. Buffer region could be 3", or $2\frac{7}{8}$ " if placed in tubing. Void and sample could be $2\frac{7}{8} \times 2\frac{7}{8} \times 4$ " long. 12 ea 1x5x10 uranium slabs needed.

10 ea $\frac{1}{8} \times 5 \times 10$
 10 ea $\frac{1}{8} \times 5 \times 10$
 2 ea $\frac{7}{8} \times 3 \times 10$
 2 ea $\frac{7}{8} \times 2 \times 10$
 2 ea $\frac{1}{8} \times 3 \times 10$
 2 ea $\frac{1}{4} \times 2 \times 10$

other slabs available

4 ea $\frac{1}{2} \times 5 \times 5$
 2 ea $\left\{ \begin{array}{l} \frac{7}{8} \times 5 \times 5 \\ \frac{1}{8} \times 5 \times 5 \end{array} \right\}$
 2 ea $\frac{1}{2} \times 5 \times 10$
 2 ea $\left\{ \begin{array}{l} \frac{1}{2} \times 3 \times 10 \\ 2 \times 10 \end{array} \right\}$



Keno calculations of this geometry surrounded by various steel reflectors would be appropriate to guide in the construction or assembly of a critical experiment. Keno code must be changed to get flux output before it will be useful!!

An alternate would be a $9\frac{1}{8} \times 9\frac{1}{8} \times 10$ surrounded by $\frac{7}{8}$ " enriched Uranium metal with more steel for reflector to achieve criticality. O.D. would be $10\frac{7}{8} \times 10\frac{7}{8} \times 11\frac{3}{4}$, except for some missing corners. ($\frac{7}{8}$ " \rightarrow 2.22 cm and reflector savings will ^{have to} be greater than 1.1 if crit thickness is > 3.3 cm.)
(Keno calculations are also possible.)

Nov 21, 1967

Smith

Nov 24, 1. Depleted UCF_6 blocks available for filter.

2. Amisen 58 calc. for 6 region reactors.

Region	Sample	Buffer	Filter	Driver	Reflector	Control
Outer Rad. (cm)	1	2	3	4	5	6
A	VOID	VOID	VOID	1	2	VOID
B	"	"	"	"	"	2
C	"	4	"	"	"	"
D	4	"	"	"	"	"
E	VOID	"	2	"	"	"
F	4	"	"	"	"	"
G	VOID	"	3	"	"	"
H	4	"	"	"	"	"
I	VOID	"	4	"	"	"
J	4	"	"	"	"	"

Mat 1	$235-IR = 4.463 E-02$	Mat 2	$Fe = 8.476 E-02$
	$238-IR = 0.324 E-02$		
3	$235-IR = 6.0342 E-02$	Mat 4	$U235-IR = 0.2356 E-02$
	$238-IR = 4.7226 E-02$		$238-IR = 4.5238 E-02$

March 25, 68 Problem of $k_{\infty} = 1$ for UF_6

Exponential column measurements could be made in existing UF_6 cylinders as a function of enrichment to determine when $B_m^2 = 0$. This concept was discussed with C. Neuman & A. Mallet but it is a large undertaking -- not enough pressure yet, but the work needs to be done.

An A-30 shipping container, modified to include several ^{diametral} radial ventrals tubes and one axial tube, should be filled with 4.5 wt% UF_6 . The Radial flux $\phi(r) = A J_0(B_r, r)$ and the z flux $\phi(z) = C e^{-\delta z}$ away from boundaries will define

$$B_m^2 = B_r^2 - \delta^2$$

If $B_m^2 = 0$, $k_{\infty} = 1$

$$B_m^2 > 0 \quad k_{\infty} > 1$$

$$B_m^2 < 0 \quad k_{\infty} < 1$$

On basis of these measurements next enrichment can be chosen for measurements. Repeating this step the enrichment for $B_m^2 = 0$ can be found

L. W. Thompson

March 25, 1968

On Friday, discussed with Dixon Callahan and Charles Crome an experimental program for 15 wt% ^{235}U . A production problem involves ^{minimum} 10 wt% ^{235}U but to get that, critical requires an estimated 400 kg or 4000 kg of U. At 15 wt% the estimate is 200 kg or 1333 kg of U, still a sizeable amount.

Buckling calculations

$$\text{Total U} = 1333.3 \text{ kg}$$

$$\text{Vol} = 1333.3 / 18.7 = 71.301 \text{ liters}$$

$$R^3 = 17022$$

$$R = 25.7 \text{ cm}$$

Assume Extrap Dist $S = 3 \text{ cm}$

$$B^2 (\text{sphere}) = \left(\frac{\pi}{28.7} \right)^2 = .10946^2 = .011982$$

Cube Size -

$$B_x^2 = .011982 / 3 = .003994$$

$$B_x = .06320$$

$$\tilde{x} = 49.7 \text{ cm}$$

$$2S = 6$$

$$43.7 \text{ cm} = 17.2 \text{ in}$$

For 18 x 18 base area

$$B_x^2 + B_y^2 = 2 \left(\frac{\pi}{45.72 + 6} \right)^2 = 2 \times .06074^2 = 2 \times .003689$$

$$B_z^2 = B^2 - B_x^2 - B_y^2 = .004604 \quad B_z = .06785$$

$$\tilde{z} = 46.30 \text{ cm} \quad z = 40.3 \text{ cm} \quad (15.86)$$

For 18 x 24 in base area

$$B_3^2 = .006092$$

$$B_3 = .07805$$

$$\tilde{z} = 40.25$$

$$\frac{25 \quad 6}{34.25}$$

$$B_3^2 = \frac{3.1416}{60.96 + 6} \Big)^2 = .04692$$

$$= .002201$$

$$B_x = \frac{3689}{005890}$$

$$005890$$

(13.25 in)

For 24 x 24 in base area

$$B_3^2 = \frac{011982}{4402}$$

$$= .007580$$

$$B_3 = 08706$$

$$\tilde{z} = 36.09$$

$$\frac{30.09}{6} \rightarrow (11.85 \text{ in.})$$

For Reflected Assemblies

$$\Delta H \quad 93\% \sim 2.5 \text{ cm}$$

$$\Delta H \quad 45 \sim 3.5$$

Assume 4.0 cm

$$S + \Delta H = 7.0 \text{ cm}$$

Sphere $B^2 = .011982 \quad \tilde{R} = 28.7 \quad R = 21.7$

$$\text{Vol} = 42.802_{47} \text{ L}$$

$$\text{Mass} = 800.4 \text{ kg}$$

Cube size = 43.7 cm - 4 cm = 39.7 cm, Vol = 62.571 L

$$\text{Mass} = 1170 \text{ kg}$$

Dr. Magmasan Mar 25, 1968

For Base Area 12 x 12 — 25 = 14 cm

$$B_x^2 + B_y^2 = \left(\frac{3.1416}{30.48 + 14} \right)^2 = 2 \times (.07063)^2 = 2 \times .004989$$

$$\begin{array}{r} .011982 \\ \hline .9978 \end{array}$$

$$B_z^2 = .002004$$

$$\bar{z} = 70.17$$

$$B_z = .04477$$

$$z = 56.17$$

(22.1 in.)

For Base Area 12 x 18

$$B_y^2 = \left(\frac{3.1416}{45.72} \right)^2 = .05261^2 = .002768$$

$$\begin{array}{r} .4989 \\ \hline .007757 \end{array}$$

$$B_z^2 = .004225$$

$$\bar{z} = 48.33$$

$$B_z = .06500$$

$$z = 34.33$$

(13.5 in.)

For Base Area 18 x 18

$$B_z^2 = \begin{array}{r} .011982 \\ .5536 \\ \hline .06446 \end{array}$$

$$B_z = 0.8029$$

$$\bar{z} = 39.13 \text{ cm}$$

$$z = 25.13 \text{ cm}$$

(9.89 in.)

Summary

	Bare Sphere	10.12 in	
	" Cube	17.2 in	
①	15.86 x 18 x 18 in	(bars)	~12
②	13.25 x 18 x 24 in	(bars)	~12
③	11.85 x 24 x 24 in	(bars)	~14
④	22.1 x 12 x 12 in	reflected	~12
⑤	13.5 x 18 x 18 in	"	~14
⑥	9.9 x 18 x 18 in	"	~14

Calculation of required $\frac{1}{2}$ in. thick pieces

	ht	Leav	12 x 12	6 x 12	6 x 6
①	$17\frac{1}{2}$	35	35	70	35
②	$15\frac{1}{2}$	31	62	62	
③	$13\frac{1}{2}$	27	108		
④	$25\frac{1}{2}$	51		102	
⑤	$15\frac{1}{2}$	31		93	
⑥	$13\frac{1}{2}$	27	27	54	27

Total required plus added contingency

$\frac{1}{2}$	58	80	40
$\frac{1}{4}$	2	3	2
$\frac{1}{8}$	2	3	2
$\frac{1}{16}$	2	3	2

Note for ~~the~~ ^{all} pieces 2 ea 6 x 6 = 1 ea 6 x 12

2 ea 6 x 12 = 1 ea 12 x 12

and this substitution can be made.

Also 2 ea $\frac{1}{4}$ can be subst. for $\frac{1}{2}$ in. thick.

Nom $\frac{1}{2} \times 12 \times 12$ pieces weighs 48.55 lbs.

or 22.064 kg

WCT set the specifications

$\pm .005$ on rectangular Dim.

$\pm .002$ on thickness

Flat and Parallel within $\pm .004$ TIR on large pcs

$\pm .002$ TIR on 6 x 6 sizes

Suggested Experimental Program for 15 wt% metal

1. Unreflected Critical Mass

- a) for an area of 18×18 in.
- b) for " " " 18×24 in.
- c) " " " " 24×24 in.

2. Reflected Critical Mass

- | | | |
|--|---|--|
| <ol style="list-style-type: none"> a) Polyethylene b) Plexiglass c) Graphite d) Depleted Uranium e) Stainless steel | } | for base dimensions of
(1) 12×12 in.
(2) 12×18 in.
(3) 18×18 in. |
|--|---|--|

3. Moderated experiments, layer experiments
reflected and unreflected

- a) Polyethylene
- b) Plexiglass
- c) Graphite
- d) Teflon

4) Additional Experiments on many of the
above

- a) Rossi- α
- b) Pulsed neutron
- c) Delayed neutron fraction
- d) Fission ratios
- e) Neutron spectra
- f) Resonance absorption in ^{238}U
- g) Flux shapes for Buckling approx.
- h) Danger coefficients and reactivity worths.

Jan 25 1968 DLM

Plutonium Critical Experiments

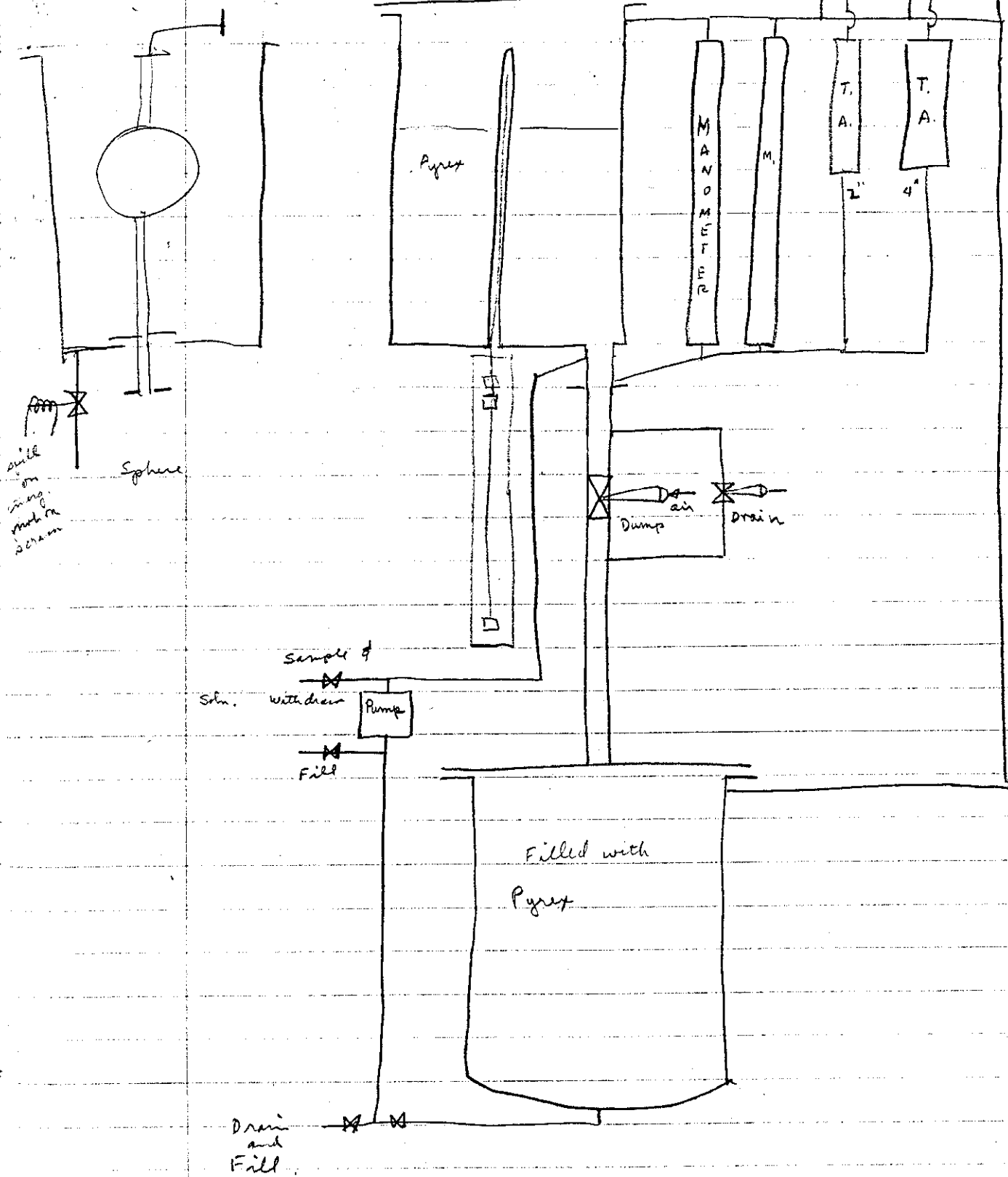
Mar 14, 1969

Poisoned solutions and Pyrex tube fixed poisons and
gadolinium - stainless steel fixed poison experiments

- Use 20 in. i.d. SS sphere made for 5% solution exp.
for boron - Cd - Pu solutions.
- Use Pyrex tubing in 20 in i.d. Cylinder 6 ft high (Pyrex
4 ft)
- Gd-Steel can be made up to 5% without seriously altering
corrosion and rolling prop. (Bu. of Mines Report WCT)

Possibly one can use TURF cell for critical experiment
Design of a simple solution system.

- 3 ft i.d. x 4 ft hi storage tank gasketed with
glass tubes 50% glass. Webster calculates
30% subcritical up to 450 g/l with no
²⁴⁰Pu.
- Metering pump having flow capability from
0.2 l/min to 20 l/min. No leakage
when stopped. Positive disp and suction. ^{Positive indication} rate setting at
a minimum _{all times!}
- 2 air operated valves + ~~2~~ manual valves.
- 2 each liquid level manometers, remote reading
- 2 each Tad Address, 2 in. od and 4 in. od., fins coated.
- 1 each 4 ft stroke counter, traversing for exp. col. flux meas.
-



$$\text{For } 30\% \text{ glass } \left(\frac{.3}{.907} - 1 \right) = - \left(\frac{n_1}{n_2} \right)^2$$

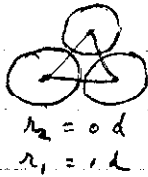
$$\left(\frac{n_1}{n_2} \right)^2 = .6693 \quad \frac{n_1}{n_2} = .8184$$

$$.8184 n_2 = n_1 \quad n_2 = 1.222 n_1$$

$$n_2 - n_1 = .199 n_2 = .2347 n_1$$

~~If glass fraction = .3; $\left(\frac{n_1}{n_2} \right)^2 = .6693$; $\frac{n_1}{n_2} = .8184$~~

Pyrex Tubing



$$\text{glass Volume} = 3 \times \frac{1}{6} \times \pi (r_2^2 - r_1^2)$$

$$\text{Total Volume} = 2 r_2 \times .866 \times (2 r_2) \times \frac{1}{2}$$

$$\text{glass Fraction} = \frac{\frac{\pi}{2} (r_2^2 - r_1^2)}{1.732 r_2^2} = .907 \left(1 - \frac{r_1^2}{r_2^2}\right)$$

$$\text{for } 50\% \text{ glass, } \frac{.5}{.907} = 1 - \left(\frac{r_1}{r_2}\right)^2, \left(\frac{r_1}{r_2}\right)^2 = .4487$$

.5513

$$\frac{r_1}{r_2} = .67$$

For various std pyrex tubing

od	wall	id	$\frac{r_1}{r_2}$	$()^2$	$1 - (r_1/r_2)^2$	glass Fract
54	2.4	49.2	.9111	.830	.17	.154
35	2.0	31	.8857	.785	.215	.195
30	1.8	26.4	.8800	.774	.226	.205
22	1.5	19.	.8636	.746	.254	.230

Roschig Rings have 40% nat. Boron.

If glass density is $\sim 2.6 \text{ g/cm}^3$

$$N_S = \frac{.04 \times 2.6 \times .6025 \times 755}{10.82} = 4.37 \text{ cm}^{-1}$$

$$\text{if } N_S = 1, \text{ then } \frac{1}{N_S} = 0.229 \text{ cm}$$

or 2.3 mm

Assume glass rods radius r_1 within contact

$$\frac{\text{glass}}{\text{Total}} = \frac{\frac{1}{2} \pi r_1^2}{1.732 r_2^2} = .907 \left(\frac{r_1}{r_2}\right)^2$$

$$\text{if glass fract} = 0.5 \quad (r_1/r_2)^2 = .5513$$

$$d_1/d_2 \quad r_1/r_2 = .7425$$

$$d_2 = 1.347 d_1, \quad d_2 - d_1 = .347 d_1, \quad \frac{d_2 - d_1}{d_2} = .258$$

For sheets of glass spacing = thickness for 50%

	spacing	thickness	glass
50% glass	1	1	1
33% "	2	1	1
25% "	3	1	1

For sheets of perforated glass

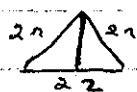
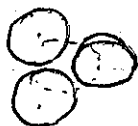
Area Removed	spacing	thickness	glass	
$\frac{1}{4}$	1	1	37.5%	glass
$\frac{1}{4}$	2	1	$\frac{1}{3} - \frac{1}{12} = \frac{3}{12} = 25\%$	glass
$\frac{1}{4}$	3	1	$\frac{1}{4} \times \frac{3}{4} = \frac{3}{16} = 18.75\%$	glass

Telephone conv. with JPN on 3-15.

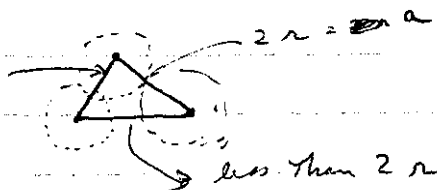
1. Pu solutions will be stored at conc. less than 200 g/l.
 ∴ the single datum of Schuckert is adequate and no more work needs to be done with Reacting rings or Pyrex glass tubes.
2. The interest in poisoned solutions does not exist, but it is in poisoned slurries in the H:Pu range of 10 - 50 where the Pu critical mass begins to decrease, i.e. there is interest in data which proves that there is a poison concentration which keeps the critical mass high, ^{200-100 kg, perhaps.} for all values of moderation ^{perhaps 10.} There is also need for similar data for mixed oxides of Pu & U.
 These data are related to dissolve criticality problems!
3. Interest remains in the poisoned pulsed column. Willing to support 235 exp. if Pu experiments cannot be set up quickly and cheaply.

It has been reported that PuO_2 microspheres do not spread much contamination. Therefore, $(\text{Pu} + \text{O})\text{O}_2$ microspheres will spread even less. Cost permitting, microsphere-paraffin compacts could be used or microspheres- $(\text{CH}_2)_n$

For close packed spheres layer separations =



$$= \begin{matrix} \text{less than} \\ 2r \\ = 1.732 r \end{matrix}$$



$$S = \frac{1}{2}(a+b+c) = \frac{1}{2}(5.464)r$$

$$S = 2.732 r$$

$$S-a = 0.732 r$$

$$S-b = r$$

$$S-c = r$$

$$2.0000 r^4$$

$$\sin A = \frac{2}{bc} \sqrt{S(S-a)(S-b)(S-c)}$$

$$= \frac{2 \times 1.4142 r^2}{3.000 r^2}$$

$$= 0.94280$$

altitude of Δ and layer separation

$$= \cancel{2r} \cdot 1.732 r \times 0.9428 = 1.6329 r$$

$$\text{Area of hexagon} = 2.59808 l^2 \quad (l = \text{side})$$

$$r = 0.8660 l, \quad l = 1.1547 r$$

$$\text{Area} = 3.4653 r^2$$

$$l^2 = 1.3333 r^2$$

$$HT = 1.6329 r$$

$$\text{Vol} = 5.6585 r^3$$

$$\frac{4.1858 r^3}{5.6585 r^3} = 0.74027$$

$$\text{Void} = 26.0\%$$

For perfect Spheres ~ 26% void volume

Assume PuO_2 25% UO_2 75%						(10x10x10) cm ³ block = 1 liter
	$ln(CH_2)_n$ <i>oxid solids</i>		$PuO_2 \rightarrow$ $CH_2 \rightarrow$	0.2548 0.790	} $\frac{H}{P_u} = 3.10$	
		<i>vol fraction</i>	<i>Vol Ratio</i> $\frac{CH_2}{PuO_2}$	$\times 3.1$		
CH ₂	PuO_2 UO_2	PuO_2		H:Pu	Pu density kg/liter	
30	70	.175	1.714	5.31		
35	65	.1625	2.154	6.68		
40	60	.15	2.667	8.26	1.62	61.7
45	55	.1375	'			
50	50	.125	4.0	12.4	1.35	
55	45	.1125	4.89	15.2		
60	40	.100	6.0	18.6	1.08	92.6
65	35	.0875	7.43	23.0		
70	30	.075	9.33	28.9	0.81	123.5
75	25	.0625	12.0	37.2		
80	20	.05	16.0	49.6	0.54	185.2
85	15	.0375	22.67	70.3		
90	10	.025	36.0	111.6	0.27	370.4

Void volumes in sphere compacts of the order of 50% will
give H:Pu ratios for $\left[\frac{3}{4}\text{UO}_2 - \frac{1}{4}\text{PuO}_2\right] - (\text{CH}_2)_n$ mixtures
of 12.4. Pu density = 1.35 g/cm³

Meeting with ADC JPN & DWM

JPN outlined problems of interest for crit. exp. with Pu.

1. For mixed oxides of variable ^{240}Pu , crit. mass data are needed for H/Pu ratios from 20-100. Solutions not possible. Effect of poisons should be evaluated in H/Pu range of < 100 so that mass values will be larger than dry oxide.
2. Pulsed column design problem -- effect of 5d-55t plates in column for increasing critical diameters. solution concentrations less than 120 g/l. Possibly substitute ^{235}U experiments if Pu is impossible for checking calculation methods.
3. $\text{UO}_2 - \text{PuO}_2$ rod lattices. high and low ^{240}Pu content. (Dissolver problem), in H/Pu range similar to (1). Does lumping reduce critical mass?
4. More data for pure plutonium in low moderation range for high ^{240}Pu content. H/Pu ~ 10 to 50 for 11.46% ^{240}Pu (Hanford has completed 5, and 15 for 2.2 and 8.0% ^{240}Pu)

ADC and JPN agreed that the proper way to proceed was to file form 189a or at least prepare them prior to informal discussions with ORNL management and/or AEC Washington (Nannum - LMFBP reactor physics.)

ADC & JPN believe that a simple contamination control system can be installed in one of the cells at 9213 suggestions include a large tank as a glove box which is ^{final} primary containment. I am not clear on how it works. I still prefer the TURF facility modified or rather with the installation of the crit exp. and instrument & control.

Both methods will be looked at!

Note that both solutions, plastic compacts and rod lattices are involved in the experimental program.

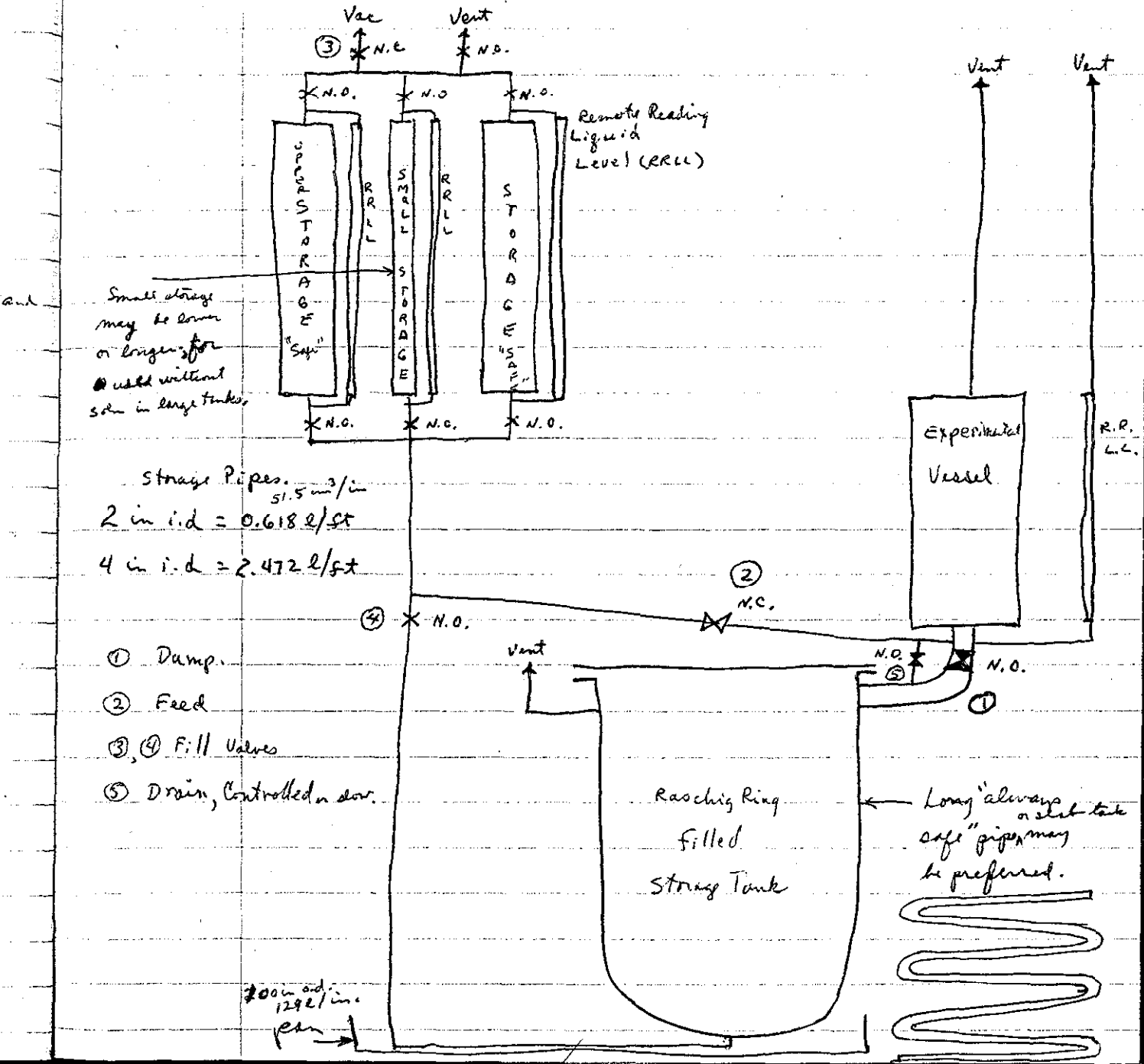
Mar 25, 69

Later conversation with JPN allowed that to get started, solution experiments may include

1. gd poisoned plates, Pa and Pa+U
2. Soluble poisons, Pa and Pa+U
3. Raschig Rings and glass tubes, Pa and Pa+U

March 26, 1969 *AWM*

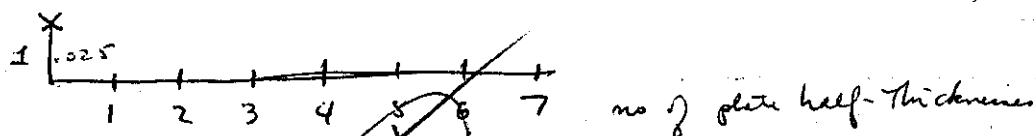
Concept of pumpless system for handling liquids -
 Vacuum lifts to storage vessels, selective "drain" to experimental vessel



October 28, 1969

Summary of thoughts on using
 Alpha Radiography as a means of measuring
 Power Distributions in HFIR Critical Experiment.

- ① Estimated Resolution from a Point Source
 or response of film relative to the response at the point



1	$1+1 = 2$	$\frac{1}{2} = 0.50$	
2	$1+4 = 5$	$\frac{1}{5} = 0.20$	
3	$1+9 = 10$	$\frac{1}{10} = 0.10$	
4	$1+16 = 17$	$\frac{1}{17} = 0.05882$	
5	$1+25 = 26$	$\frac{1}{26} = 0.03846$	
6	$1+36 = 37$	$\frac{1}{37} = 0.02703$	
7	$1+49 = 50$	$\frac{1}{50} = 0.02000$	
8			
9			
10		$\frac{1}{101} \approx 0.01$	

- ② Film Blackening proportional to Absorbed Dose.

therefore β dose should be principal dose from
 fission products in an irradiated fuel plate.

Most film ^{at speed} when exposed to IR should be in mid range of density.
 E. G. Qupton

③ In discussions with Fitzgall, Y-12 Radiographers using careful calibration, reproducibility, or errors could be reduced to 2-5%. Confirmed by E. Gyton ORNL A.P. (gave me Type M-2 Industrial X-RAY Film (Kodak) to try.)

④ Micro densitometer at Y-12,
 Max scan speed 5 cm / MIN (too slow)
 Max slit size 0.05×2 mm. (too small)
 Max plate length ≈ 25 cm (too short)
 Max plate width ≈ 25 cm adequate
 Surely there are others to do the job at hand.

⑤ Assume $\frac{1}{x^2}$ law for intensity at a point

Assume uniform plate, infinite in x & y

Diagram: A horizontal line represents a source. A vertical line represents a film. A point 'o' is marked on the film. A distance 'x' is indicated from the source to the film. A small circle with a dot is shown on the film, representing a point source.

$$\int_0^{\infty} \int_0^{2\pi} k d\theta \left[\frac{dx}{1+x^2} \right] = 2\pi \int_0^{\infty} \frac{x dx}{1+x^2} = -\frac{2\pi}{2} \left[\frac{-2x dx}{1+x^2} \right]$$

$$= -\pi \left[\frac{1}{1+x^2} \right]_0^{\infty} = \pi$$

Intensity due to infinite plate

Intensity at a point o due to immediate area having radius R

Done

$$\frac{-\pi \left[\int_0^R \frac{-2x dx}{(1+x^2)} \right]}{\pi} = \frac{\pi \left(1 - \frac{1}{1+R^2} \right)}{\pi} = \frac{R^2}{1+R^2}$$

	$R \times .025$ inches	Intensity Fraction		of source has circular hole
1	.025	1/2	0.50	unit rel. .50
2	.05	4/5	0.80	rel x2 .20
3	.075	9/10	0.90	x3 .10
4	.100	16/17	0.9412	x4 .0588
5	.125	25/26	0.9615	x5 .0385

96? The response of a point detector to an infinite planar source is due to 5% diff of detector from plane source.

This resolution should be adequate for the flux measurements and power dial meas.

as well as the uranium distribution.

For the latter, highest film speed + long exposure times are expected to be required.

DM

234

U decay Assume 0.01 fraction

Assume 50 mg sample / cm²

$$\frac{dN}{dt} = N\lambda = 1.25 \times 10^{-6} \times 10^{24} \text{ atoms} \times \frac{2.196 \times 10^{-8}}{2.48 \times 10^5 \text{ years}}$$

$$= 0.885_{48} \times 10^{-13} \times 1.25 \times 10^{18} = 1.107 \times 10^5 \frac{\alpha}{\text{sec}}$$

from 50 mg

Decay scheme \rightarrow 47 keV γ 26%gamma decay = 2.878×10^4 (47 keV γ) / sec 235

U decay assume 100% and 50 mg sample

$$N = 50 \times 10^{-3} \times 2.563 \times 10^{23} \text{ atoms/gm} = 1.2815 \times 10^{22} \text{ at.}$$

$$\lambda = \frac{2.196 \times 10^{-8}}{7.13 \times 10^8 \text{ y}} = .308 \times 10^{-16}$$

$$\frac{dN}{dt} = N\lambda = 3.947 \times 10^3 \text{ d/sec}$$

Decay scheme is complicated - not given
in Old Reactor Physics Handbook -Assume $\sim 1 \gamma$ per α , $\sim 4 \times 10^3$ α / sec from ^{235}U

If so ^{234}U has largest contribution to low energy
counting and dose at film surface!

GWT

Jim Henry - Y-12

Auto radiograph of ^{90}Sr point source
gave spot $\sim 2\text{ mm}$ in diameter, 2 mc strength

$$2 \times 3.7 \times 10^{-3} \times 10^{10} = 7.4 \times 10^7 \beta / \text{sec}$$

$$\text{Geom Factor} = 2.5$$

$$\sim \frac{3 \times 10^7 \beta}{\pi 0.1^2} \sim 1 \times 10^9 \beta / \text{cm}^2$$

HFIR CE #3

$$171 \text{ inner plates } 171 \times 20 \times 3.0 \times 6.45_{16} = 66,193 \text{ cm}^2$$

$$369 \text{ outer } " \quad 369 \times 20 \times 2.65 \times " = 126,174 "$$

$$1.92367 \times 10^5 \text{ cm}^2$$

$$\frac{1}{A} = 5.2 \times 10^{-6}$$

If critical exp = 1 watt for 1200 sec (20')

$$3.1 \times 10^{10} \times 1.2 \times 10^3 \times 10 \beta / \text{fission} = 4 \times 10^{14} \text{ total } \beta$$

$$4 \times 5.2 \times 10^{14} \times 10^{-6} = 2.08 \times 10^9 \beta / \text{cm}^2 \text{ delayed}$$

$$\text{or } 2.08 \times 10^8 \text{ fissions/cm}^2$$

If fuel plate has 50 mg/cm^2 and 2.08×10^8 fissions/cm²

$$\rho \sigma N t = 2.08 \times 10^8$$

$$\rho = 580 \times 10^{-24} \text{ cm}^2$$

$$N = 1.25 \times 10^{-4} \times 10^{24}$$

$$t = 1200$$

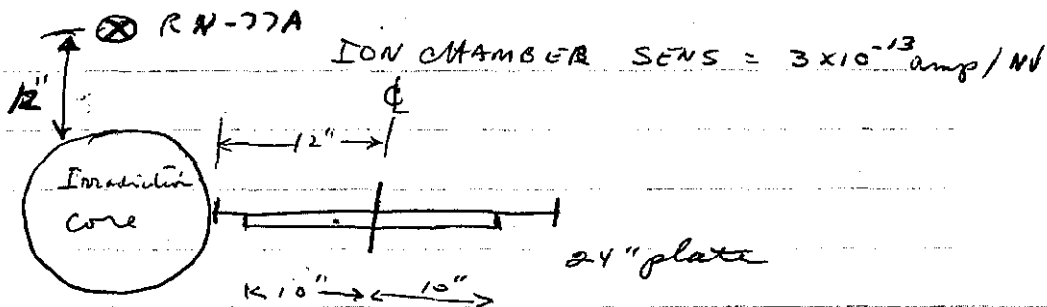
$$\rho = \frac{2.08 \times 10^8}{5.80 \times 10^2 \cdot 1.25 \times 10^{-4} \times 1.2 \times 10^3}$$

$$= 2.2 \times 10^6 \text{ n/cm}^2 \text{ sec}$$

DWM

Preliminary Irradiation Plan

85



$$3 \times 10^{-13} \times 2.2 \times 10^6 = 6.6 \times 10^{-7} \text{ amp.}$$

MSM

Oct 31, 1969

Estimate of dose from X-ray absorption in water
(Curve does not go down to 47 keV - APEX 1716)

$$\begin{aligned} \mu &= \sim 20\% \text{ per cm at } \frac{9.4}{4.7} \text{ keV per cm} \\ &\times 2.88 \times 10^4 \text{ r/cm}^2 \\ &\frac{2.7}{1.35} \times 10^3 \text{ dN/cm}^3 \times 1.59 \times 10^{-12} \text{ erg/keV} \\ &= \frac{4.32}{2.7} \times 10^{-4} \text{ erg/cm}^3 \\ \text{Exp. Time for } 100 \text{ erg/cm}^3 &\rightarrow \frac{2.32}{.86 \times 10^5} \text{ sec.} \\ &= \frac{268}{.86} \text{ days} \\ &= \frac{268}{.86} \text{ days} \end{aligned}$$

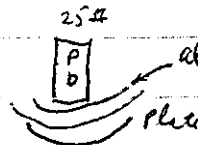
~~Assumption~~ If Mass Abs Coeff is 1.0 (Most elements inc. rapidly at low energy) then Exp. time is 54 days!

EXP #1 Film

@ 11³⁵ AM Oct 31, 1969 Film in contact with inner plate

I

@ 11⁵⁵ AM Nov 10, 1969 Film removed from plate
240.5 hours.



10-31-69

Inner Annulus Fuel Plate Irradiation

(No foils in HFIRCE #3 Removable

Plate # IE-BA

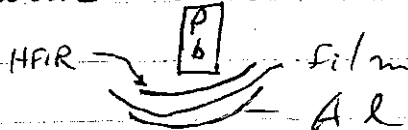
| Plate ~ Vert at
midplane

Log

OK-2

19 Slug (3.85) in \square array corner Missing
 K-1 @ 12" K-2 @ 19 1/2 in. Log @ 18 in.

On period Scram by faulty Inst
 Recovered and back up on longer period

2⁴² @ 1×10^{-7} on K-1 @Level at 2.4×10^{-7} , Drop to 2.0×10^{-7} 3¹² Shut down Assembly~3²⁰ Centie Pie read 5n at hot end.3²⁵ Film #2 in contact with Plate dummy Plate # Pb weights4²⁵ Remove Film #2, 60 MIN EXPOSURE ~1R at hot end.4³⁰ Film #3 in contact with plate > 13' exp.4⁴³ Film #3 removed.

DW

November ~~13~~ ³

Films #2 & 3 developed by Fitzgerald & Smith

#2 Hot end very black, #3 underexposed some detail on hot end only.
Cold end very light

Ideal exposure greater than 2×10^6 m/sec/cm² at mid point for 30 minutes.

A halo appears in side of each Foil hole
explanation is no 10 mil aluminum attenuation of the radiation from the sandwich.



Film appears mottled when the o.d. is ideal

- a) Due to non uniform uranium dist.
- b) Due to non uniform 10 mil thickness
- c) Due to non uniform paper thickness.

Nov 11, 1969

a fuel scan with Metal & Ceramics X-Ray Absorption fuel scanner showed fuel inhomogeneities similar to the variations in power density.

b) and c) most likely, do not apply.

DW Magnuson

exposure

weights

d.

X-10 Inspection Engineering or Metallurgy group
 of Radiographers
 Discussions with Dr. K. Klindt & _____ ?
 KLM

Factors which affect reproducibility of Film

Optical Density can be reproduced to ± 0.02 by

- a) Developing in Fresh Developer
- b) Using film all from the same batch
and recalibrating each batch.
- c) Developing film from same run all
at the same time.
- d) Independent of film type.
- e) Type M, Type AA, Medical No Screen
and Type K are all high contrast film
Factor film becomes "grainy"
- f) Type AA $\sim 4 \times$ faster than M
Medical No Screen $\sim 16 \times$ faster
Type K faster yet but is definitely
grainy.
- g) Calibration should probably be done
with same spectrum; however
it will be a 2nd order effect for small
changes in film from batch to batch.
- h) Recommend setting up the
developing at 9213, and use
fresh developer for each batch of
10-15 films constituting one

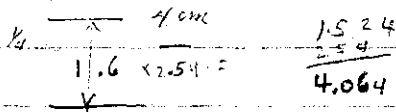
irradiation and exposure.

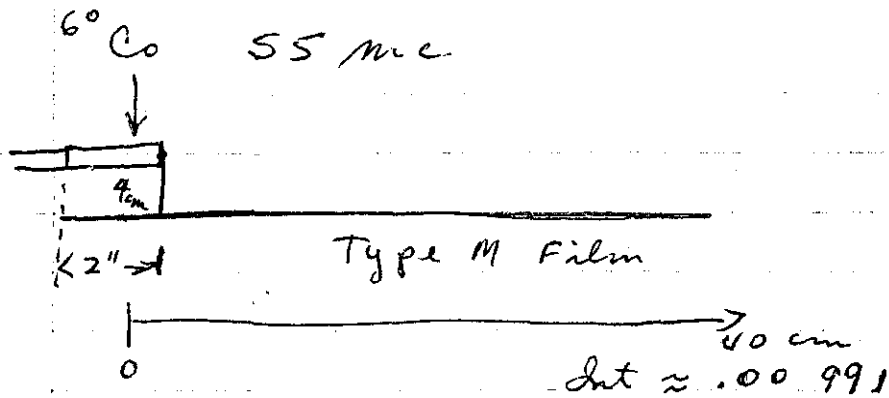
Since Medical No Screen is 16x faster, it would be best to use Type K probably is too grainy, but would be useful for auto-radiographing the uranium activity in unmediated plates.

Optimal Exposure for Calibration of Type K

55 mc ⁶⁰Co Source 86 mR/HR @ 1 meter
 8,600 R/HR @ 10 cm
 34.4 R/HR @ 5 cm

R ²	R ² /R ²	40cm	x 4.5625 = 53.75	4cm
0	100	1.0	2.0000	0
7.2		.64	1.806	3
10	200	.5	1.69897	4 cm
20	500	.2	1.30103	6
30	1000	.1	1.0	8 cm
40	17	.0588	0.76938	12 cm
50	26	.038	0.57978	
60	37	.027	0.43136	
70	50	.020	0.30103	1.30103
80	64	.01538		
90	82	.0122		
100	101	.009901		0



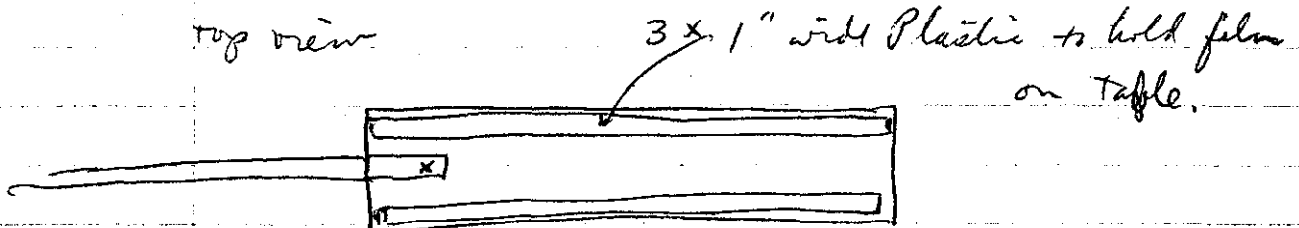


11:51:40 Start Irradiation

Film #4. 1 hr 30'

1:21:40

top view



Second film #5

Start 1:24 PM } 1 hr Exposure.
 Stop 2:24 PM

Re Telephone Calvin Hopper

#15 24 mc Two pellet - radiograph shows 1!
 '63 40 mc
 #20 55 mc Only one.
 '63 181 mc - radiograph shows 2!

Assume Opt Density at mark of 3.6

Type M @ 5' development

$$O.D. \ 3.4 \quad \log \text{ rel exp} = 2.81$$

$$O.D. = 3.0 \quad \log \text{ rel exp} = 2.7$$

$$\text{slope} = \frac{0.6}{0.1} = 6$$

$$\Delta O.D. = 0.02$$

$$\Delta \log \text{ Rel exp} = .1/6 = 0.01667$$

$$\text{Rel exp} = 1.039$$

Type AA.

$$O.D. \ 3.57 \quad \log \text{ rel exp.} = 2.20$$

$$\underline{2.95} \quad \underline{2.10}$$

$$\frac{.62}{.10} = 6.2 \quad \text{slope}$$

Med. No Screen

$$3.5 \quad \sim 1.75$$

$$3.0 \quad \sim 1.64$$

$$\frac{2.81}{1.75}$$

$$\approx 1.06$$

$$\approx 11.48$$

Type K

Most sensitive or fastest but grainy ^{x faster}

Film Buyer 3-4203 Wilson

Eastman Kodak Repres. for Ind. X-ray

Tom Gilliland 584-5393

7752 Evanshire Dr.

all
have
approx

same
slope

GAF sold Micro densitometer rights to
Technical Operations, Inc Snow Hill, 1969

Nov 11, 1969

Previous irradiation of fuel plate

30' @ 2×10^{-7} amp on K-2 @ 12" from coreEstimated current required for optimum irradiation
which was perhaps 6" from core

$$\sim \frac{2 \times 10^{-5} \text{ amp}}{3 \times 10^{-13} \text{ amp/mv}} = 0.67 \times 10^8 \text{ mv}$$

If in critical experiment chamber is placed

where flux is 0.67×10^7 (1/10 of core)

$$\text{current} = 0.67 \times 10^7 \times 3 \times 10^{-13} = 2 \times 10^{-6}$$

If Chamber is $7\frac{1}{2}$ " from core factor is 20.

$$\frac{0.67 \times 10^7}{20} = 0.33 \times 10^6$$

$$\times 3 \times 10^{-13} = 1.0 \times 10^{-7}$$

∴ Find outside location where flux is down
by a factor of ~ 100 so that chamber current
can be $\sim 1.0 \times 10^{-9}$, a decade below saturation.

Nov 11, 1969 Re Phone

34203 Wilson purchasing

Ready Pack Type M $4\frac{1}{2} \times 24$ "

75 sheets per box

min order 3 boxes @ \$31.73 per box

45 day Delivery

Nov 18, 1969

- Jesse Erwin -- Filler sides of inner plates for #FIRCE #2 plates were badly blistered. Therefore
- | | | |
|---|------|------------------------------|
| 1 | U-9 | convex side is filler side |
| 2 | T-7 | and concave is fuel side. |
| 3 | U-11 | Films 7 & 9 should be denser |
| 4 | U-13 | than 6 & 8, respectively. |
| 5 | U-10 | |

Eddy current method could detect differences in plate sides, but lacking a standard no interpretation could be made. (D odd of Metallurgy)

Nov-12, 1969

HFIR CE #2 in Tank at West End Reed, & Coors

Critical water height determined prior to irradiation exp. $\sim 2\frac{1}{2}$ in H_2O above fuel

After 10 min at critical plate read 25 mR on Cutie Pie for 10 min decay. #4 plate used

Log N hanging outside tank. not moved. $= 1.1 \times 10^{-10}$

other slots empty in room

Plate #6 (U-10) interchanged 30' Irr @ $\log N = 10$.

Will go to $\log N = 10$ if Bldg alarms will permit. OK

Start Irr @ $10^{\frac{45}{15}}$ @ 3.7 m $\log N > 20'$
 stop Irr @ $11^{\frac{15}{15}}$

Expose Film #6 and #7 Start @ 11²⁷



Stop @ 1⁰⁰ PM

Plate read ~ 2.5 R contact with cutie Pie

only 0.600 R/HR thru 50 mil of Al as mounted in film holder.

Start #8 bottom } irradi @ 1⁰⁰ PM } 19 $\frac{1}{2}$ hrs
 #9 top } 8³⁰ AM }

60 mR/HR thru 50 mil of Al

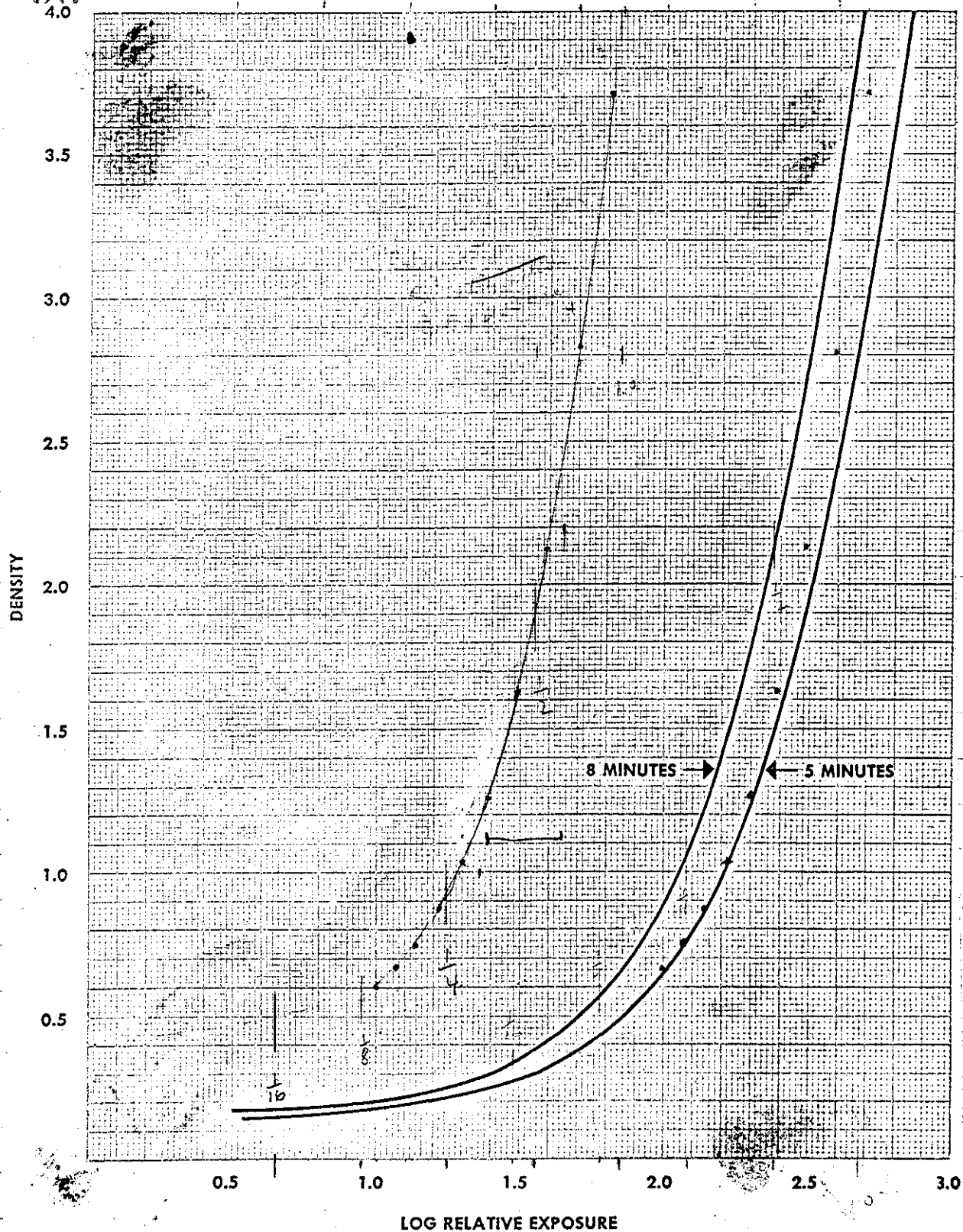


Fig. 87—Characteristic curves of Kodak Industrial X-ray Film, Type M, for direct or lead foil screen exposures to x-rays or gamma rays. (Kodak Rapid X-ray Developer or Kodak Liquid X-ray Developer and Replenisher at 68 F.)

Inner
Mid
Outer

Nov 19, 1969

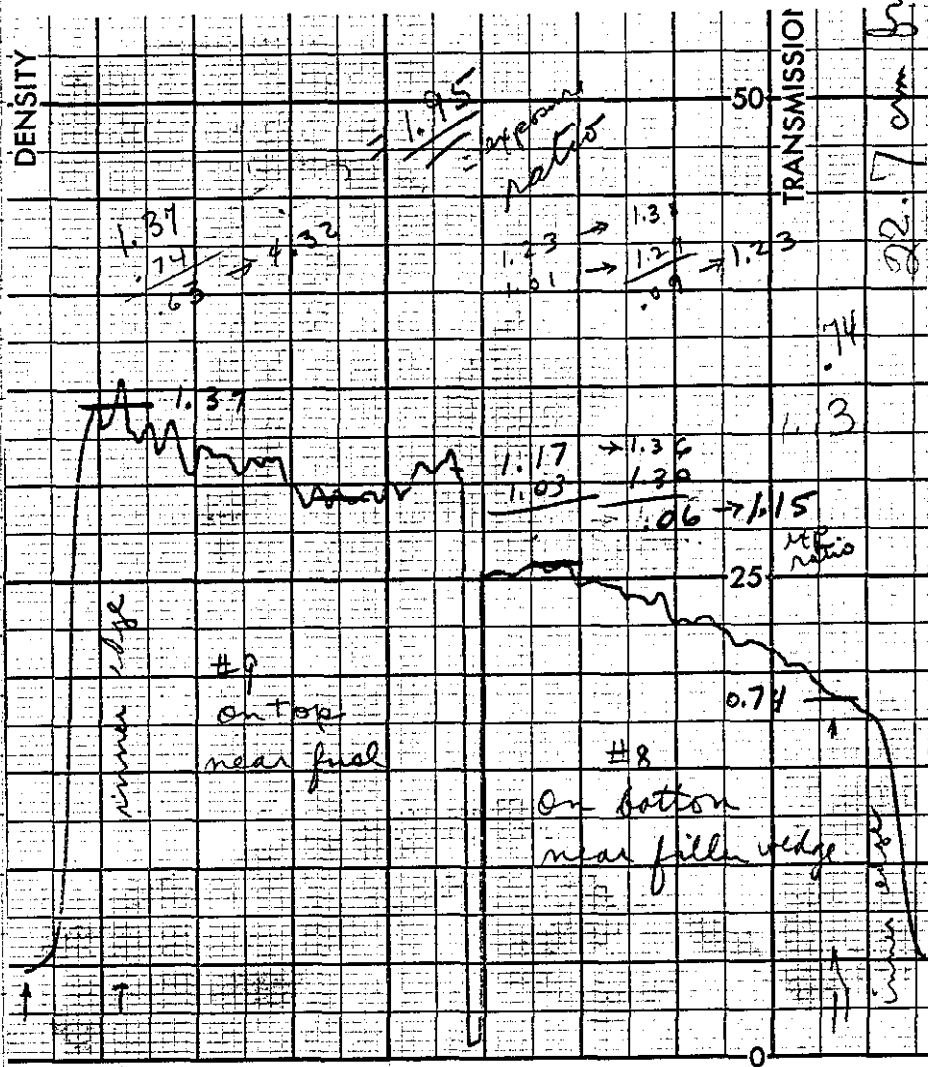
(Hacker)

Observed #9 film has greater O.D. than film #8

Densitometer trace
between 8 and 9

- a) Difference is greatest on inside edge where filler wedge is thickest, thereby attenuating exposure to #8. Same is also closer to No 9.
- b. Difference between 8 and 9 is smallest where filler wedge is thinnest - - near plate middle.

Exp Ratio
Inner 4.32
Mid 1.15
Outer 1.23



Exp Ratio 4.32

MP ratio

Nov 19

Discussions with T.O. Gilliland, Eastman Kodak, Nov 13, 69
reaffirmed the following

- a) Film response should be reproducible to 0.01 - 0.02 O.D.
- b) Medical No Screen film, which is ~16X faster than Type M, will not show grain structure
- c) Attenuation of filler strips and the location of the fuel strips will give rise to different responses at various positions across the plate and a film calibration (or response) curve will have to be determined for each axial section^{or strip} of the fuel plate.
- d) Automated film developer should give better results than hand developing.
- e) Calibration with different energy sources should be the same for O.D. vs log relative exp. that is by normalizing at one point the calibration curves should have same shape!
will check with Kodak Technical representative!
- f) He offered to supply me with some Med No Screen for some experiments.
- g) Medical No Screen should be available in any of the sizes that Type M is available.
i.e. a 24 in. ready pack could be supplied.
also as a 200 ft roll (or longer)
- h) Report - Autoradiography Spec Train Div

ORINS June 15-27, 1953 Overman Boyd Smith
DUM

Dec 11, 1969

M. Sims discussed with WCT & DWM locations of points on fuel plates where information is needed for input to the heat transfer code, a total of 203 points per plate. Included are measurements of the power distribution if the uranium is located above the top or bottom of the core.

Ways of getting this information were discussed

- a) shifting normal plates up or down in order to get fuel above or below nominal core
- b) making an extra long plate from two plates.

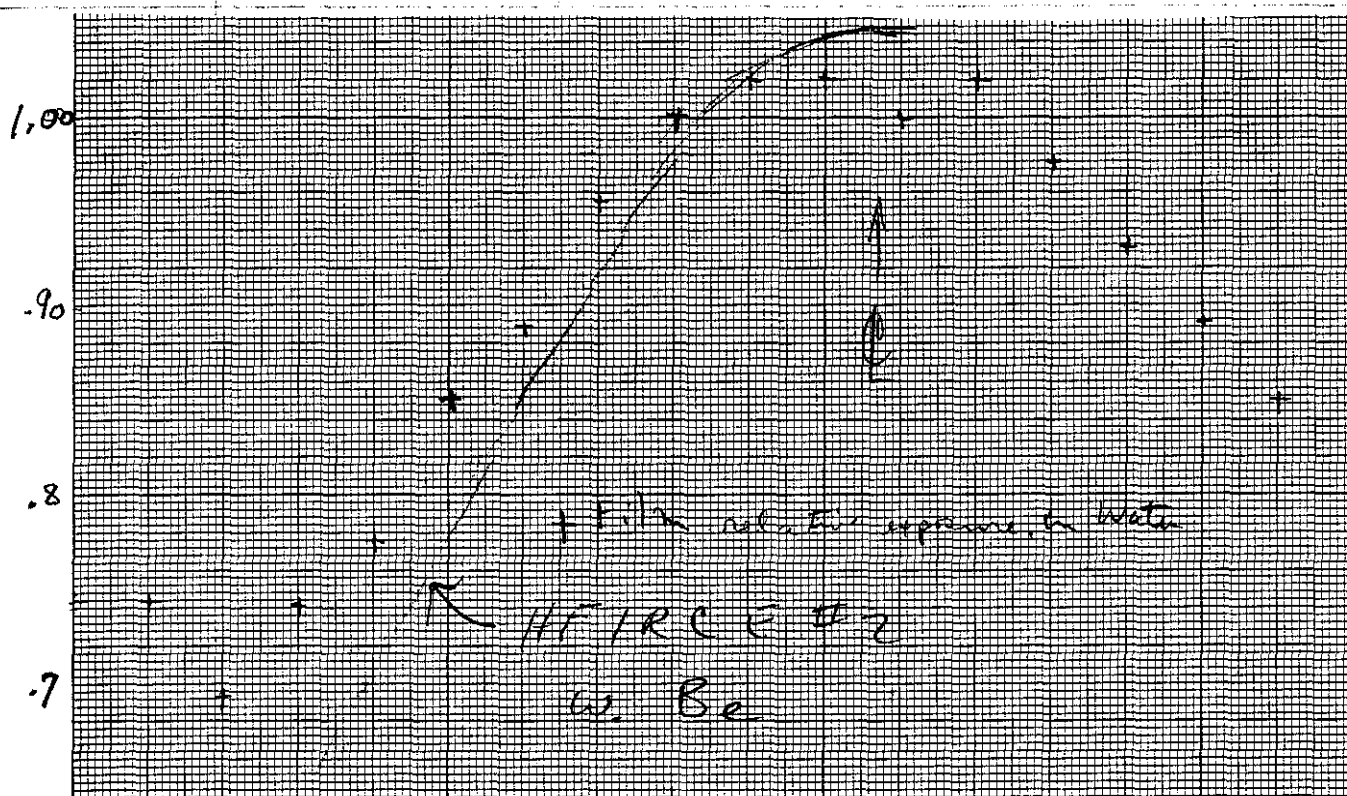
The latter is probably preferred, one method would be to take two plates cut them to the proper length - weld them together, ^{making a uranium section ~ 21" long} form into the proper shape - - decide later on preferred method a) or b).

Ralph Turner of Eastman Kodak pointed out that NoScreen could not be developed in the Y-12 automatic film processor -- therefore will use Type A

December 11, 1969

Borrowed (from M. Silverman), Macbeth Quantalog
 TD-100 A Densitometer with 2 mm aperture
 (1 & 3 mm lost?)

Built table around it covered with
 graph paper so that film edge could be
 moved by increments easily in X and Y
 direction thru scanning plate easily.
 Using calibration from Kodak, Radiograph
 in Modern Industry, an axial distribution
 was measured from the film exposed in
 HFIR CE #2 without Beryllium and
 compared to orig crit exp results.



Dec 11, 1969

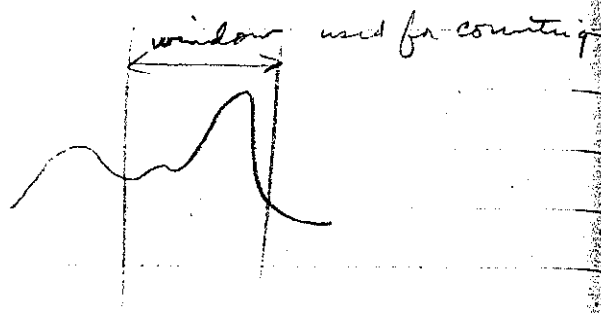
Recheck of count rate for ^{235}U in SA-2
 for ~ 50 mg Sample 5' 167 789
 Bkg 730

$$167\,059/5 = 33,400 \frac{\text{C}}{\text{M}}$$

Baseline 0.55

Upper level 0.60

Window x1

Memory $\frac{1}{4}$ 

Foil area of $\frac{1}{8} \times \frac{3}{2}$ in = 1.21 cm^2
 contain a minimum of $25.5 \text{ mg } ^{235}\text{U}$

$\frac{1}{4}$ in o.d. foil area = $.3167 \text{ cm}^2$

minimum of ~ 6.4 mg

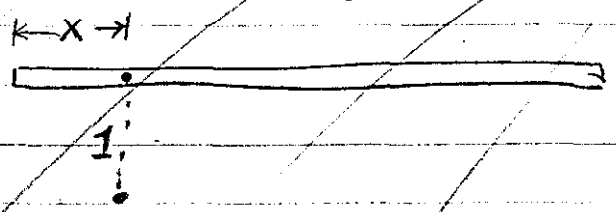
$$\frac{6.4}{50} = .128 \sim \frac{1}{8} \quad \text{or} \sim 4000 \text{ C/MIN}$$

N.B. Scintillators could be adjusted to be closer
 to the sample thereby increasing count rate

$\frac{7}{16}$ in foil \rightarrow area = $.4948 \text{ cm}^2$

High background on SA-1, ¹³⁷Cs peak
 Disconnect lower scintillator - no peak
 Disconnect upper " - peak remains
 Perhaps ¹³⁷Cs has been "spilled" on
 lower scintillator -- should be
 investigated.

Calculated response at edge of fuel for line source!



Film response

$$\int_0^x \frac{dx}{(1+x^2)} + \int_0^{\infty} \frac{dx}{(1+x^2)} = \tan^{-1} x \Big|_0^x + \frac{\pi}{2}$$

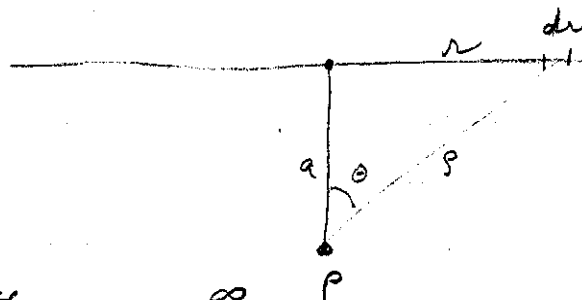
$$\frac{2 \int_0^{\infty} \frac{dx}{(1+x^2)} = \pi}{\pi} = 1.5708$$

x	$\tan^{-1} x$	$\times 1/180$	$+ .5$
1	45°	.25	.75
2	$63^\circ 26'$.3529	.8524
3	$71^\circ 34'$.3976	.8976
4	$75^\circ 58'$.4221	.9221
5	$78^\circ 41'$.4372	.9372
6	$80^\circ 32'$.4474	.9474
7	$81^\circ 52'$.4548	.9548
10	$84^\circ 18'$.4683	.9683

Response at edge of fuel is 50%
 at 5 x the source distance inside, the response is
 93.7% of the value for infinite plane source.

December 12, 1969

Finite & infinite plane source



$$\rho = \sqrt{a^2 + r^2}$$

S = source strength
 per unit area

$$\cos \theta = \frac{a}{\rho}$$

Infinite source

$$\begin{aligned} \text{Flux at P} &= \int_0^{\infty} \frac{S \times 2\pi r dr}{4\pi \rho^2} \times \frac{a}{\rho} \\ &= \frac{aS}{2} \int_0^{\infty} \frac{r dr}{(a^2 + r^2)^{3/2}} \\ &= \frac{aS}{2} \left[\frac{-1}{(a^2 + r^2)^{1/2}} \right]_0^{\infty} = \frac{S}{2} \end{aligned}$$

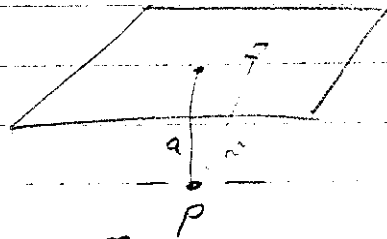
Finite Source

$$\begin{aligned} &= \int_0^R \frac{aS}{2} \left[\frac{-1}{a(1 + R^2/a^2)^{1/2}} + \frac{1}{a} \right] \\ &= \frac{S}{2} \left[1 - \frac{1}{(1 + R^2/a^2)^{1/2}} \right] \end{aligned}$$

Infinite Source with hole

$$= \int_R^{\infty} \frac{S}{2} \times \frac{1}{(1 + R^2/a^2)^{1/2}}$$

Finite and Infinite Plane source, Rect geometry



Infinite source

Flux at P = $\int_{-\infty}^{\infty} dx \int_{-\infty}^{\infty} dy \frac{S dy}{4\pi(a^2+x^2+y^2)} \times \frac{a}{(a^2+x^2+y^2)^{1/2}}$

= $\frac{aS}{\pi} \int_0^{\infty} dx \int_0^{\infty} \frac{dy}{(b^2+y^2)^{3/2}}$ where $b^2 = a^2+x^2$

= $\frac{aS}{\pi} \int_0^{\infty} dx \left[\frac{y}{b^2(b^2+y^2)^{1/2}} \right]_0^{\infty}$ $\sim \frac{1}{b(b^2+y^2)^{1/2}}$
when $y=0 \rightarrow 0$

= $\frac{aS}{\pi} \int_0^{\infty} dx \times \frac{1}{(a^2+x^2)} = \frac{aS}{\pi} \left[\frac{1}{a} \tan^{-1} \frac{x}{a} \right]_0^{\infty}$ $y=\infty \rightarrow \frac{1}{b^2}$

= $\frac{S}{\pi} \times \frac{\pi}{2} = \frac{S}{2}$

$x=0$
 $\tan^{-1} 0 = 0$
 $x=\infty$
 $\tan^{-1} \infty = \frac{\pi}{2}$

no glab
 hence
 side

Finite source $x = X$ width X
 = $\frac{aS}{\pi} \left[\frac{1}{a} \tan^{-1} \frac{X}{a} \right] = S \frac{\tan^{-1} \frac{X}{a}}{\pi}$

$\int_{-\infty}^{\infty} \frac{S dy}{4\pi(a^2+y^2+a^2)} = \frac{S}{\pi} \int_0^{\infty} \frac{dy}{(b^2+y^2)}$ where $b^2 = a^2+x^2$
 $= \frac{S}{\pi} \int_0^{\infty} \frac{dy}{(a^2+x^2+y^2)^{1/2}}$ $b = (a^2+x^2)^{1/2}$
 $= \frac{S}{\pi} \int_0^{\infty} \frac{dy}{(a^2+x^2+y^2)^{1/2}} = \frac{S}{2} \left\{ \log \left[x + (a^2+x^2)^{1/2} \right] \right\}$ $z=0 \rightarrow \tan^{-1} 0 = 0$
 $x=\infty \rightarrow \tan^{-1} \infty = \frac{\pi}{2}$
 = $\infty \times \left[1 + \left(1 + \frac{a^2}{x^2} \right)^{1/2} \right]$

R/a	Circular Source	Hole in Inf. Plane
$1/10$.0050	.9950
$1/5$.0194	.9806
$1/4$.0299	.9701
$1/3$.0413	.9587
$1/2$.1056	.8944
1	.2929	.7071
2	.5528	.4472
3	.6838	.3162
4	.7675	.2425
5	.8039	.1961
10	.9005	.0995

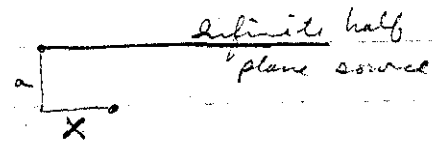
~~$$g = \frac{S \times 2\pi}{4\pi} \int_0^{\infty} \frac{r dr}{(a^2 + r^2)} = \frac{S}{2} \left[\frac{1}{a} \tan^{-1} \frac{r}{a} \right]_0^{\infty}$$

$$r=0 \quad \tan^{-1} 0$$

$$r=\infty \quad \tan^{-1} = \frac{\pi}{2}$$

$$= \frac{S\pi}{4a}$$~~

Relative Flux near Plane edge



X/a	\tan^{-1}	Relative Flux
-20		.0159
-15		.0211
-10		.0317
-5		.0628
-4		.0771
-3		.1074
-2	$-63^{\circ}26$.1426
-1	-45	.2500
0	0	.5000
1	45° .5000	.7500
2	$63^{\circ}26$.8524
3	$71^{\circ}34$.8976
4	$75^{\circ}58$.9221
5	$78^{\circ}41$.9372
10	$84^{\circ}18$.8430 .4683	.9683
15	$86^{\circ}11$.8618	.9789
20	$87^{\circ}8$.8713	.9841

If film is $\approx .035$ in from source of radiation, then at a dist of .175 in. (5a)
 from edge the film response is 93.7% of an inf plane.

Febr 26, 1970

Auto Radiograph of Source No 20 55mc of Co^{60}
 showed that there were two pellets in the source
 one of which was 2.2 x the other. The stronger
 source was located $1/2$ in. from the tip, the
 weaker source $1 1/16$ in from the end. 1 in.

According to Calvin Hopper, the source
 No 20 55mc Co^{60} should be only one pellet.

Film No 10

Source #18, according to his records, 28mc
 was comprised of two pellets. Using
 the Cutie Pie, the two source strengths
 are correct and labels have not been switched.

Film # 11 30 sec exposure with No #18 source ^{28mc}
 in contact to locate source in holder.

5x7 type AA. Film - Source $1 5/16$ " from Tip

Weaver -- No latent image fading between
 exposure and developing for life of
 film! Therefore wait time before developing
 is unimportant

©9204-4

Densitometer Measurements on F10
 1 minute irradiation on Type M film

0.0	1.24	0.61
	<u>.17</u>	<u>.17</u>
	1.07	.44
	<u>.44</u>	

.34 → 2.2 X ratio of source strengths

$\frac{55}{3.2}$	=	$\frac{17.1875}{3.4375}$ mc	17mc
	x2	34.375	
	x1.2	<u>37.8125</u> mc	<u>38mc</u>

30 sec Irradiation with 28mc source on Type AA

O.D = 2.49 → 1.65

light fog = .7

net density ~ 2.3

AA Film is $\frac{60}{30} \times \frac{55}{28} \times 1.6$ faster than M

$\frac{\log}{\text{Rel Exp.}}$ = 6.29 X faster than M

2.49 1.65

Est OD

1.42 1.44

Relative speed Kodak

$\frac{AA}{M} \frac{100}{15} = 6.67$

in sources
 in 20 (55mc)
 were not
 pellet

.21



$$g = \frac{S}{4\pi R^2} = \frac{S}{4\pi} \frac{1}{a^2 + x^2} = \frac{S}{4\pi a^2} \frac{1}{1 + (x/a)^2}$$

x	$(x/a)^2$	$1 + (x/a)^2$	$\frac{1}{1 + (x/a)^2}$	$\log_{10} \frac{1}{1 + (x/a)^2}$
0	0	1	1	2
a	1	2	0.50	1.699
2a	4	5	0.20	1.301
3a	9	10	0.10	1.000
4a	16	17	0.05882	.7695
5a	25	26	0.03846	.5850
6a	36	37	0.02703	.4319
7a	49	50	0.02	.301
8a	64	65	0.015385	.1871
9a	81	82	0.012195	.0862
10a	100	101	0.009901	-.9957

March 2, 1970

Set $a = 3.125$

set $x = 0$ 1 in from edge of film #12

set $x = 8\frac{3}{4}$ in at edge of film #13

Irradiate film for 60 minutes (Elect Timer

1 38

Start

2 38

Stop

March 30

Exposed film with 3 CF sources

B

= 1.7m Dian
120000 c/s

A

0.5 30000 c/s

C

0.5 8000 c/s

June 1, 71

Quantalogs returned to M. S. Sorenson (several?)
months ago.

W. J. M.